# **AEROSPACE ENGINEERING (LM52)**

(Brindisi - Università degli Studi)

Teaching SPACE PROPULSION MOD. 2	<b>Teaching in italian</b> SPACE PROPULSION <b>Course year</b> 1 MOD. 2	
	<b>Teaching</b> SPACE PROPULSION MOD. 2	Language INGLESE
GenCod A003310	SSD code ING-IND/07	Curriculum Percorso comune
<b>Owner professor</b> Maria Grazia DE GIORGI	<b>Reference course</b> AEROSPACE ENGINEERING	
	Course type Laurea Magistrale	Location Brindisi
	Credits 6.0	Semester Secondo-Semestre
	<b>Teaching hours</b> Ore-Attivita-frontale: 54.0	Exam type Orale
	For enrolled in 2020/2021	Assessment
	Taught in 2020/2021	<b>Course timetable</b> https://easyroom.unisalento.it/Orario

BRIEF COURSE DESCRIPTION	This course presents aerospace propulsive devices with particular focus on rocket engine
REQUIREMENTS	-Fluid dynamic and fluid machinery
COURSE AIMS	<ol> <li>Gain knowledge of different types of aero-engines (turbojets, turbofans, ramjets) and to understand the aerodynamic and thermodynamic characteristics of major rocket components.</li> <li>Develop the knowledge and skills to analytically and numerically solve problems related to aerospace propulsion systems.</li> <li>Develop skills in working independently.</li> <li>Develop skills in critical evaluation of scientific literature.</li> <li>Develop skills in planning and presentation of scientific talks and reports.</li> </ol>
TEACHING METHODOLOGY	Theory and practical activities
ASSESSMENT TYPE	The final exam consist of two part: 1)Written and oral examination covering all material covered in course 2)assignments and individual project



## FULL SYLLABUS

*Rocket Nozzles and Thrust* Performance and nozzle design. Convective Heat Transfer

#### Combustion and Thermochemistry

Perfect gas law and thermodynamics review,equilibrium Thermochemistry, adiabatic flame temperature calculations, non-Equilibrium Flows. Rocket nozzle thermochemistry.

Solid Rocket Motors

General description, interior ballistics, component design goals and constraints.

#### Liquid Rocket Motors

General description, engine cycles, power balance calculations, component design fundamentals. Combustion of Liquid Propellants ; Injection and Mixing ; Stability; Pressurization and Pump Cycles; Turbomachinery Performance

Trajectory Analysis and staging

The rocket equation, vertical trajectories, multistage rockets.

*Electric Propulsion* General description and classification of electric propulsion systems, performance analysis.

*Hybrid rockets* Classification, Challenges, and Advantages of Hybrids

### REFERENCE TEXT BOOKS

• Aerothermodynamics of Gas Turbine and Rocket Propulsion Gordon C. Oates eISBN: 978-1-60086-134-5 print ISBN: 978-1-56347-241-1 DOI: 10.2514/4.861345

• Hill, P., and Peterson, C., Mechanics and Thermodynamics of Propulsion, Addison-Wesley Publishing Co., 1992,

• George P. Sutton, Oscar Biblarz, Rocket Propulsion Elements, 7th Edition John-Wiley & Sons, Ltd., ISBN: 0-471-32642-9

Course note

