The course is aimed at introducing the student to the methods for modeling the dynamic behavior of an aircraft as a function of its aerodynamic configuration, propulsion system and inertial characteristics. Based on models derived on first principles, the students will learn the tools necessary for the determination of aircraft characteristics in terms of static and dynamic stability and response to controls. The course is focused on the dynamics of rigid aircraft. Effects of structural deformation on stability and control are introduced at an elementary level. A few notions on rotorcraft dynamics (helicopter trim and rotor blade flapping dynamics) and satellite attitude dynamics and control are also provided.

Tutorials will allow the students to apply the notions learned to representative examples and case studies, maturing the capability of interpreting aircraft and spacecraft motion as a function of controls.

Basic knowledge of fluid-dynamics and a good knowledge of flight mechanics and analytical dynamics are highly recommended.

A the end of the course the student is expected to be able to
1) determine trim conditions, aircraft stability and response to controls for conventional configurations;
2) understand basic features of rotary wing aircraft dynamics and its response to controls;
3) understand basic features of rigid spacecraft dynamics and how to control it;
4) handle mathematical and numerical tools for simulating aircraft and spacecraft dynamic behavior.
TEACHING METHODOLOGY

The course is delivered with class and laboratory activities, in three different forms:

- **standard class lectures**, where the teacher presents methods and models; students are encouraged to participate by discussing validity of the assumptions at the basis of the models and physical meanings of the results derived from the analysis performed; **example: derive the expression of aircraft neutral point**;

- **tutorial classes**, during which problems are stated, where the students refine their understanding, by numerically evaluating aircraft performance from geometric, propulsion and aerodynamics characteristics; the teacher supports the class by recalling relevant models and highlighting the procedure; some calculations (e.g. for a different set of parameters) can be proposed to the students as homework; **example: evaluate the position of aircraft neutral point from aircraft geometric and aerodynamic data**;

- **computer lab. classes**, where students are required to write simple computer programs for performing parametric analysis, and/or use or implement Simulink models for simulation; **example: evaluate aircraft response in simulation for different control inputs**.

Results from homework and computer lab classes will be collected in a report to be delivered and discussed during the oral exam.

ASSESSMENT TYPE

The exam is oral.

The exam starts with a discussion of the projects proposed during the tutorials and lab. classes in order to evaluate the capability of the student in analyzing complex problems, where numerical tools or a large number of calculations are required, using some mathematical programming software and/or simulation tools.

The oral exam also includes the discussion of more general aspects regarding aircraft and helicopter dynamics, spacecraft attitude dynamics and control.

ASSESSMENT SESSIONS

Exam diets are performed according to current University regulations (3 exam diets at the end of each semester, 1 exam diet in September, 2 extraordinaty exam diets for students who finished the regular course).

Exact dates are provided on the University website, as soon as they are available.

OTHER USEFUL INFORMATION

**Orario di ricevimento:** al termine delle lezioni, oppure previo appuntamento da concordare via e-mail (indirizzo istituzionale giulio.avanzini@unisalento.it).

**Office hours:** at the end of the lectures or arranging a meeting, to be scheduled by sending a request via e-mail to giulio.avanzini@unisalento.it.
FULL SYLLABUS

- Equations of motion for rigid aircraft (4 hours).
- Equilibrium in the longitudinal plane: longitudinal static stability; longitudinal control and trim; directional stability and dihedral effect; lateral-directional control; non-symmetric flight (6 hours).
- Tutorials on trim curves and static stability (4 hours)
- Dynamic stability: linearization of aircraft equations of motion; stability derivatives; longitudinal dynamics; lateral-directional dynamics (16 hours)
- Tutorials on dynamic stability and response to controls (4 hours)
- Nonlinear phenomena: inertial coupling; autorotation; spin (2 hours).
- Rotary-wing aircraft: helicopter commands; swashplate; flap dynamics (4 hours).
- Project 1: Laboratory on basic facts in aircraft flight simulation (4 hours)
- Rigid spacecraft dynamics: free-spinning motion and passive stabilization (4 hours).
- Rigid spacecraft active control: sensor and actuators; control techniques (4 hours).
- Project 2: Laboratory on spacecraft attitude dynamics simulation (4 hours)

REFERENCE TEXT BOOKS

**Flight Dynamics**


*In Italiano*


**Suggested readings from...**


**Handbooks on spacecraft attitude dynamics and control**


*In Italiano*